

United States of America
Department of Transportation -- Federal Aviation Administration

Supplemental Type Certificate

Number SA01640CH

This certificate issued to Elliott Aviation Inc.
Quad City Int'l Airport
P.O. Box 100
Moline, IL 61266-0100

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part 23 of the Federal Aviation Regulations. (See Type Certificate Data Sheet A24CE as amended by Special Condition No. 23-115-SC, dated April 2, 2002, Docket No. CE181 for complete certification Basis)

Original Product - Type Certificate Number : A24CE
Make : Hawker Beechcraft Corp.
Model : 200, 200C, 200CT, 300, 300LW, B200
B200C, B200CT, B200T, B300, B300C

Description of Type Design Change:

Installation of a Universal Avionics Systems Corp. (UASC) EFI-550 Electronic Flight Instrument System (EFIS) in accordance with Star Aviation Master Data List (MDL) No. R34-0488, Rev. IR, dated January 09, 2009, or later FAA approved revision.

Limitations and Conditions:

1. FAA Approved Airplane Flight Manual Supplements (AFMS), for the Installation of a UASC EFI-550 EFIS (2 Display Option), Star Aviation Document No. R34-0499, Rev. IR, dated February 02, 2009, or later FAA approved revision, is a required part of this STC.
2. FAA Approved AFMS, for the Installation of a UASC EFI-550 EFIS (4 Display Option), Star Aviation Document No. R34-0500, Rev. IR, dated February 02, 2009, or later FAA approved revision, is a required part of this STC.

(see continuation page)

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application : December 28, 2001

Date reissued : August 20, 2009

Date of issuance : May 15, 2002

Date amended : February 02, 2009



By direction of the Administrator

Kenneth M. Webb
For (Signature)

Melvin D. Taylor, Manager,
Atlanta Aircraft Certification Office

(Title)

United States of America
Department of Transportation - Federal Aviation Administration
Supplemental Type Certificate
(Continuation Sheet)

Number SA01640CH

Date reissued: August 20, 2009
September 08, 2008

Date amended: February 02, 2009
January 27, 2006
August 12, 2003

Limitations and Conditions (continued):

3. AFMS, Elliott Aviation Report No. FS2006-450, Rev. 1, FAA approved August 2, 2003, or Star Aviation Document No. R34-0499, Rev. IR, dated February 02, 2009 (following STC amendment dated Feb. 02, 2009), or later FAA approved revision, is required onboard aircraft modified in accordance with Configuration (-02) of Star Aviation MDL No. R34-0488, Rev. IR, dated January 09, 2009, or later FAA approved revision.
4. AFMS, Elliott Aviation Report No. FS2006-451, Rev. 1, FAA approved August 2, 2003, or Star Aviation Document No. R34-0500, Rev. IR, dated February 02, 2009 (following STC amendment dated Feb. 02, 2009), or later FAA approved revision, is required onboard aircraft modified in accordance with Configuration (-04) of Star Aviation MDL No. R34-0488, Rev. IR, dated January 09, 2009, or later FAA approved revision.
5. AFMS, Elliott Aviation Report No. FS2006-452, Rev. 1, FAA approved August 2, 2003, or later FAA approved revision, is required onboard aircraft that have the Collins FCS-65H Automatic Flight Control System installed and are modified in accordance with Star Aviation MDL No. R34-0488, Rev. IR, dated January 09, 2009, or later FAA approved revision.
6. AFMS, Elliott Aviation Report No. FS2006-499, Rev. 1, FAA approved January 27, 2007, or later FAA approved revision, is required onboard aircraft modified in accordance with Configuration (-01) of Star Aviation MDL No. R34-0488, Rev. IR, dated January 09, 2009, or later FAA approved revision.
7. This approval should not be extended to other aircraft of this model on which other previously approved modifications are incorporated, unless it is determine by the installer that the interrelationship between this change and any other previously approved modifications will produce no adverse effect upon the airworthiness of that airplane.
8. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.

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